

Predicting Vehicle Crashworthiness: Validation of Computer Models for Functional and Hierarchical Data

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Abstract

The CRASH computer model simulates the effect of a vehicle colliding against different barrier types. If it accurately represents real vehicle crashworthiness, the computer model can be of great value in various aspects of vehicle design, such as the setting of timing of air bag releases. The goal of this study is to address the problem of validating the computer model for such design goals, based on utilizing computer model runs and experimental data from real crashes. This task is complicated by the fact that (i) the output of this model consists of smooth functional data, and (ii) certain types of collision have very limited data. We address problem (i) by extending existing Gaussian process-based methodology developed for models that produce real-valued output, and resort to Bayesian hierarchical modeling to attack problem (ii).

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Additionally, we show how to formally test if the computer model reproduces reality. (Supplemental materials for the article are available on-line.)

KEY WORDS: Air bag timing; Bayesian analysis; Bias; Hierarchical modeling; Hypothesis testing; Validation; Vehicle design.

1 INTRODUCTION

1.1 The Computer Model for Vehicle Crashworthiness

The CRASH computer model simulates the effect of a collision of a vehicle with different types of barriers. Proving ground tests with prototype vehicles must ultimately be made to meet mandated standards for crashworthiness, but the computer model plays an integral part in the design of the vehicle to assure crashworthiness before manufacturing the prototypes. How well the model performs is therefore crucial to the vehicle design process.

CRASH is implemented using a non-linear dynamic analysis (commercial) code, LS-DYNA, using a finite element representation of the vehicle. The main focus is on the velocity changes after impact at key positions on the vehicle.

Geometric representation of the vehicle and the material properties play critical roles in the behavior of the vehicle after impact and the necessary detailing of these inputs leads to very time consuming computer runs (from 1 to 5 days on a standard workstation). Obtaining field data involves crashing of full vehicles, so that field data is obviously also limited. Studying CRASH is thus inherently data-limited — both in terms of computer runs and field data — presenting a basic challenge to assessing the validity of the computer model in accurately representing real vehicle crashworthiness.

There are many variables and sources of uncertainty in the vehicle manufacturing process and proving ground test procedures that, in turn, induce uncertainties in the test results. The acceleration and velocity histories (i.e., the acceleration and

velocities over the initial 100 millisecond time interval of the crash) of two production vehicles, subjected to 30mph zero degree rigid barrier frontal impact tests, are shown in Figure 1. These test results indicate the variability inherent in the field data.

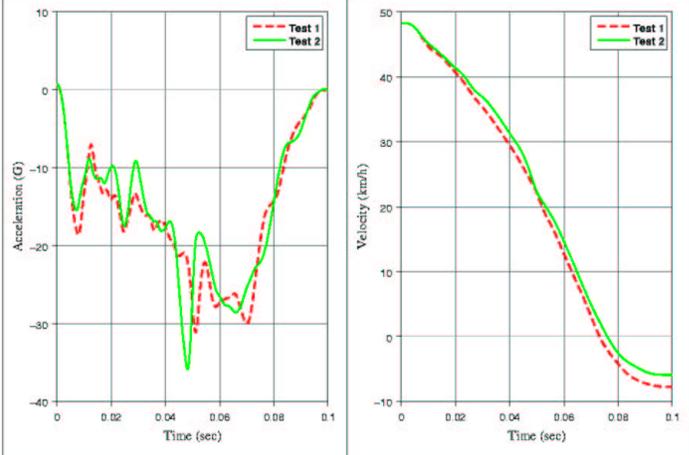


Figure 1: Acceleration and velocity pulses in the occupant compartment from 30mph zero degree rigid barrier frontal impact tests for two production vehicles.

1.2 Engineering Goals of the Study

The primary engineering motivation for the analysis herein was to help judge whether the CRASH computer model is an adequate reflection of reality, and hence can be used with confidence in vehicle design. For instance, we will see that CRASH seems to be an excellent computer model at higher velocities, but has a definite bias at lower velocities; such conclusions can give important guidance to the engineers in their efforts to improve the computer model.

We also estimate a bias for the computer model, which can be used for bias-corrected prediction. This can be particularly important when the computer model is used to analyze design questions faced by the engineers. One design question of particular interest is related to setting the firing time of air bag systems. To address this question, in Section 4.4 we focus on assessing how well the model is able to predict the velocity 30ms before the displacement reaches 125mm, which is a key quantity in

determining when an airbag should be fired (see also Section 2.4).

Our statistical analysis also involves development of an *emulator* of the computer model, which is simply a fast numerical approximation to the computer model. It will be seen that this emulator is itself accurate enough for use in the analysis of this design question, thus allowing much more intensive exploration of design space than would be possible if the computer model itself had to be run at all potential design points.

1.3 Statistical Background

General discussions of the entire Validation and Verification process for computer models can be found in Roache (1998), Oberkampf and Trucano (2000), Pilch et al. (2001), Trucano et al. (2002), and Santner et al. (2003). We focus here on the last stage of the process: that of assessing the accuracy of the computer model in predicting reality, and in using both the computer model and field data to make predictions, especially in new situations.

Because a computer model can virtually never be said to be a completely accurate representation of the real process being modeled, the relevant question is “Does the model provide predictions that are accurate enough for its intended use?” Thus predictions need to come with what were called *tolerance bounds* in Bayarri et al. (2007b), indicating the magnitude of the prediction error.

This focus on giving tolerance bounds, rather than stating a yes/no answer as to model validity, arises for three reasons. First, models rarely give highly accurate predictions over the entire range of inputs of possible interest, and it is often difficult to characterize regions of accuracy and inaccuracy; for instance, we will see that, for certain inputs, the computer model of vehicle crashworthiness yields quite accurate predictions, while for others it does not. Second, the degree of accuracy that is needed can vary from one application of the computer model to another, as will be seen in

study of the different uses of CRASH. Finally, tolerance bounds account for *model bias*, the principal symptom of model inadequacy; accuracy of the model cannot simply be represented by a variance or standard error.

The key components of the approach outlined here are the use of Gaussian process response-surface approximations to a computer model, following on work in Sacks et al. (1989), Currin et al. (1991), Welch et al. (1992), and Morris et al. (1993), and introduction of Bayesian representations of model bias and uncertainty, following on work in Kennedy and O’Hagan (2001) and Kennedy et al. (2002); Higdon et al. (2004), Lee et al. (2006), Campbell (2006), Gramacy and Lee (2006) and Gramacy and Lee (2008) are more recent references. A related approach to Bayesian analysis of computer models is that of Craig et al. (1997), Craig et al. (2001), Goldstein and Rougier (2003) and Goldstein and Rougier (2004), and Goldstein and Wooff (2007) which focus on utilization of linear Bayes methodology to address the problem.

1.4 Required Methodological Extensions

Bayarri et al. (2007b) described a general framework for validation of complex computer models and applied the framework to two examples. The extensions of this methodology that were needed to deal with the CRASH model were the following:

Problem 1 – Smooth Functional Output: In Bayarri et al. (2007b), the examples considered involved real-valued model outputs. The output of CRASH, however, is a time-dependent function. Thus, in validating the computer model with field data, one must compare functions, a more difficult enterprise.

Problem 2 – Hierarchical Modeling: A common problem in computer modeling is that codes (and field experiments) can be run under differing conditions, where the differences are not completely quantifiable and where data may be scant – or even lacking – for some of the conditions. In CRASH, this arises when treating

collisions of different barriers; there are few data for center pole collision or for right-angle collision, but there are reasonable amounts of data for straight-frontal and left-angle collision (see Table 1). Therefore, using hierarchical modeling, data from the various experiments can be combined to obtain improved predictions. In addition, the methods can be used for prediction under untried conditions, as long as the new conditions are deemed to be compatible with the hierarchical modeling assumptions, although the uncertainties associated with these predictions may be quite large.

Problem 3 – Testing Model Correctness: The validation question initially asked by many computer modelers is “Can you establish that the computer model is correct?” In Section 6 it is shown how one can formally conduct a test to answer this question. The result of the test will virtually always be that there is conclusive evidence that the computer model is not correct, but formally providing an answer to this question can be of pedagogical value.

1.5 Overview

Section 2 outlines the key elements of the problem, and discusses the basic strategy that is utilized to deal with functional data. Approximation of the computer model is considered in Section 3, while the Bayesian validation analysis is given in Section 4. Section 5 introduces the hierarchical methodology for dealing with related scenarios (differing barrier collisions in CRASH), while Section 6 considers the formal testing of model validity. Finally, Section 7 concludes with a summary of our findings for the CRASH model, and a technical summary of the methodology.

Most of the details of the computations are relegated to the Appendix and to a companion technical report, Bayarri et al. (2005), which is available on-line under “Supplementary Material” at <http://pubs.amstat.org/loi/jasa>.

2 KEY ELEMENTS OF THE PROBLEM

2.1 Inputs

The inputs to the computer model will be denoted by a vector \mathbf{x} . In the case of CRASH, the two key inputs are $x_1 = \textit{velocity at impact}$ and *barrier type*. Of course, the computer model has numerous other inputs (indeed thousands, if one counts the finite-element basis of the computer model) but, for this case-study, the desired engineering focus was in studying the computer model in terms of these two inputs.

The methodology from Bayarri et al. (2007b) that we are adopting does not accommodate qualitative inputs such as barrier type. Hence, to deal with the full range of barrier types, we will resort to hierarchical analysis. However, three of the barrier types (left angle, straight frontal, and right angle) can be converted into the quantitative input $x_2 = \textit{angle of impact}$. For most of the paper, we will analyze only the data for straight frontal collision (the most extensive data set), and will then be using \mathbf{x} to just represent *impact velocity*, but in Section 5 we also consider angle and \mathbf{x} will then represent the pair (*impact velocity, angle*).

In CRASH, the selected inputs for running the computer model and collecting field data were chosen for reasons other than conducting a model validation (and before the initiation of this work), and are given in Table 1; note that replicates exist of the field data, which is highly useful for model validation.

Each computer run or field test resulted in acceleration data curves as in the left panel of Figure 1, corresponding to various locations on the vehicle. (For the field data, these acceleration curves were from sensors placed at various locations.) The primary output of interest, and that which we shall focus on here, is the relative velocity of the “Sensing and Diagnostic Module”, SDM, situated under the driver’s seat, relative to a free-flight dummy. This relative velocity is obtained by integrating the observed SDM acceleration curves and then subtracting the impact velocity x_1

Impact velocity (km/h) used in model	barrier type	Impact velocity (km/h) of field tests
19.3	straight frontal	19.3
25.5	straight frontal	25.5
28.9	straight frontal	28.9
32.1	straight frontal	32.1
35.3	straight frontal	35.3
38.4	straight frontal	38.4
41.3	straight frontal	41.3, 41.3
49.3	straight frontal	49.4, 49.2, 49.4, 49.3, 49.3, 49.4
56.4	straight frontal	56.4
22.5	left angle	22.5
32.2	left angle	32.2
40.2	left angle	40.2, 41.4, 41.5
41.9	left angle	41.9
49.3	left angle	49.5, 49.2
56.2	left angle	56.2
57.3	left angle	57.3
28.9	right angle	28.9
31.9	right angle	31.9
41.7	right angle	41.7, 41.8
48.3	right angle	48.3
19.3	center pole	19.3
25.5	center pole	25.5
32.0	center pole	32.0
36.8	center pole	36.8
40.3	center pole	40.3
48.6	center pole	48.6

Table 1: Data is available at these inputs.

from the resulting velocity (it being assumed that the dummy maintains velocity x_1 over the time interval of interest). The resulting functions vary (at least theoretically) between 0 at the time of impact $t = 0$ and $-x_1$ at the time the vehicle is stationary.

The collection of all inputs for which the computer model was exercised will be denoted $D^M = \{\mathbf{x}^1, \dots, \mathbf{x}^L\}$. These inputs are given in Table 1. Note that L ranges from 4 to 9 depending on the barrier type (or is 20 when the angle is used to represent the first three barrier types).

The collection of all inputs to field trials (proving ground crashes of vehicles) will be denoted $D^F = \{\mathbf{x}^{*1}, \dots, \mathbf{x}^{*l}\}$, where there could be repeat inputs (i.e., replications in the field trials). These are also given in Table 1 and note that l ranges from 5 to 15 depending on the barrier type (or is 30 when the angle is used to represent the

first three barrier types).

2.2 Incorporating Functional Outputs

There are several possible approaches that can be taken to adapt the methods of Bayarri et al. (2007b) to the setting of functional data of the type depicted in the right panel of Figure 1.

One common approach is to represent the functions that arise through a basis expansion (e.g., a polynomial expansion), taking only a finite number of terms of the expansion to represent the function. The coefficients of the terms in this expansion would then be viewed as a collection of real-valued output functions, with the methodology from Bayarri et al. (2007b) being applied to each. Examples using a wavelet basis for ‘rough’ output functions can be found in Bayarri et al. (2007a). See also Higdon et al. (2008) for a related approach.

In our model, the functional data for velocity (which is the primary interest) is rather smooth, and so we turn instead to the most direct possibility, which is to combine t with \mathbf{x} , thus enlarging the input space to $\mathbf{z} = (\mathbf{x}, t)$. Since we only consider discrete input values here, we further must assume that we have only observed the function at a discrete set of points, $D^T = \{t_1, \dots, t_N\}$. If N is chosen large enough and the points at which we record the function are chosen well, then the function values at these N points will adequately represent the entire function. Similar approaches are taken in Conti and O’Hagan (2007), Rougier (2007), and McFarkand et al. (2008).

As will be seen in Section 3, it is computationally important to discretize all functional outputs (model-run and field data) at the same set of time points D^T . We have chosen $D^T = \{1, 3, 5, 7, 9, 11, 13, 15, 17, 20, 25, 30, 35, 40, 45, 50, 55, 60, 65\}$, where all measurements are in milliseconds (ms). More points were chosen in the region $t < 20ms$, since information from this region will be seen to be more important in estimating the primary quantity of interest, related to air bag firing. Times above

65 *ms* are unimportant for the same reason, and so there was no need to include such points in the design. These 19 time points provide a good compromise between adequate representation of the function (for the desired purposes) and numerical efficiency.

2.3 Data

As discussed above, the data we analyze is the relative SDM velocity in the field runs and computer model runs over the (augmented) input domains $D^F \times D^T$ and $D^M \times D^T$. (The Cartesian product form arises because D^T was chosen to be the same for each function.) These data will be denoted by $\mathbf{y}^M \equiv \{y^M(\mathbf{x}, t) : \mathbf{x} \in D^M, t \in D^T\}$ and $\mathbf{y}^F \equiv \{y^F(\mathbf{x}, t) : \mathbf{x} \in D^F, t \in D^T\}$, vectors of lengths that we henceforth denote by m and n , respectively, constructed from lexicographic ordering of the inputs.

2.4 Evaluation Criteria

Computer models are typically used for many purposes, so overall fidelity to reality of the computer model predictions is an issue of considerable interest. Hence, we will be interested in comparing the entire function predicted by the computer model with the function observed from the field data, over a wide range of inputs.

We will also focus on a more specific feature that is of engineering interest, namely the SDM velocity calculated 30*ms* before the time the SDM displacement (relative to the free-flight dummy), DISP, reaches 125*mm*. Call this quantity CRITV. The air bag takes around 30*ms* to fully deploy, which is why this particular evaluation criterion is important. Our analysis takes account of the dependence between displacement and velocity (displacement is the integral of velocity); it is a useful feature of the Bayesian approach we adopt that dealing with composite criteria, such as this, is no more difficult than dealing directly with the outputs.

3 APPROXIMATING THE COMPUTER MODEL

The CRASH model can take days to run, and a fast approximation – an *emulator* – is needed for the statistical analysis (which we perform via Markov Chain Monte Carlo simulation (MCMC), requiring thousands of computer model runs).

We denote the (now scalar) output of the CRASH model at input $\mathbf{z} = (\mathbf{x}, t)$ by $y^M(\mathbf{z})$. An expensive computer code, such as the one for the CRASH model, is only run at some few inputs (here $\mathbf{z} \in D^M \times D^T$) and should thus be viewed as an unknown function at all other values of \mathbf{z} . An emulator is an approximation $\hat{y}^M(\mathbf{z})$, which also has the desirable property of having a variance function $V^M(\mathbf{z})$ that measures the accuracy of $\hat{y}^M(\mathbf{z})$ as an estimate of $y^M(\mathbf{z})$. A response surface approach that achieves both these goals is the Gaussian process response surface approximation (GASP), described in Sacks et al. (1989) and Kennedy and O’Hagan (2001). We thus assume that $y^M(\mathbf{z})$ has a prior distribution given by a Gaussian process with mean $\mu(\cdot)$, correlation function $c^M(\cdot, \cdot)$, and unknown precision (1/variance) = λ^M , to be denoted by $\text{GP}(\mu(\cdot), \frac{1}{\lambda^M} c^M(\cdot, \cdot))$.

Process mean: There is a clear trend in the velocity output, as can be seen in the right panel of Figure 1 (although recall that we will be considering the relative velocity, found by subtracting the initial impact velocity, x_1). The relative velocity thus starts at 0 and declines to a value near $-x_1$, and hence it is reasonable to introduce a term that reflects this trend, the simplest being of the form $\mu^M x_1 t$, with μ^M being unknown.

Process correlation: We choose this to be of the form

$$c^M(\mathbf{z}, \mathbf{z}^*) = \exp\left(-\sum_{\ell=1}^d \beta_{\ell}^M |z_{\ell} - z_{\ell}^*|^{\alpha_{\ell}^M}\right). \quad (3.1)$$

Here, d is the number of coordinates in \mathbf{z} : either 2 or 3 for CRASH (with inputs augmented by t), depending on whether *angle* is included as an input. The α_ℓ^M take values in $(1, 2)$, and the β_ℓ^M in $(0, \infty)$. The product form of the correlation function (each factor is itself a correlation function in one dimension) facilitates the computations made later. Prior beliefs about the smoothness properties of the function will affect the choice of $\boldsymbol{\alpha}^M = \{\alpha_\ell^M : \ell = 1, \dots, d\}$. For example, the choice $\alpha_\ell^M = 2$ for all ℓ reflects the belief that the function is infinitely differentiable, plausible for many engineering models. We denote all the covariance parameters by $\boldsymbol{\theta}^M = (\lambda^M, \boldsymbol{\alpha}^M, \boldsymbol{\beta}^M)$, where $\boldsymbol{\beta}^M = \{\beta_\ell^M : \ell = 1, \dots, d\}$.

Finite dimensional distribution: From the Gaussian process assumption, \mathbf{y}^M , conditional on the hyperparameters, is multivariate normal with covariance matrix $\boldsymbol{\Gamma}^M = \mathbf{C}^M(D^M \times D^T, D^M \times D^T)/\lambda^M$, where $\mathbf{C}^M(D^M \times D^T, D^M \times D^T)$ is the matrix with (i, j) entry $c^M(\mathbf{z}_i, \mathbf{z}_j)$, for each pair \mathbf{z}_i and \mathbf{z}_j in $D^M \times D^T$. Once \mathbf{y}^M is observed, this yields a likelihood function for the parameters $\boldsymbol{\theta}^M$ and μ^M based solely on the observed \mathbf{y}^M .

Prediction (approximation) at a new input: Given \mathbf{y}^M and the unknown parameters μ^M and $\boldsymbol{\theta}^M$, $y^M(\cdot)$ is again a Gaussian process with mean and covariance functions given respectively by (see, for example, Stein (1999))

$$E[y^M(\mathbf{z}) \mid \mathbf{y}^M, \mu^M, \boldsymbol{\theta}^M] = \mu^M x_1 t + \mathbf{r}_z' (\boldsymbol{\Gamma}^M)^{-1} (\mathbf{y}^M - \mu^M \boldsymbol{\Psi}) \quad (3.2)$$

$$\text{Cov}[y^M(\mathbf{z}), y^M(\mathbf{z}^*) \mid \mathbf{y}^M, \mu^M, \boldsymbol{\theta}^M] = \frac{1}{\lambda^M} c^M(\mathbf{z}, \mathbf{z}^*) - \mathbf{r}_z' (\boldsymbol{\Gamma}^M)^{-1} \mathbf{r}_{z^*}, \quad (3.3)$$

where $\mathbf{r}_z' = \frac{1}{\lambda^M} (c^M(\mathbf{z}, \mathbf{z}_1), \dots, c^M(\mathbf{z}, \mathbf{z}_m))$, $\boldsymbol{\Gamma}^M$ is given above, and $\boldsymbol{\Psi}$ is the column vector consisting of the values of $x_1 t$ corresponding to the inputs in $D^M \times D^T$.

If \mathbf{z} is a new input value, the response surface approximation to $y^M(\mathbf{z})$, given $(\boldsymbol{\mu}^M, \boldsymbol{\theta}^M)$, is simply (3.2), $E[y^M(\mathbf{z}) \mid \mathbf{y}^M, \boldsymbol{\mu}^M, \boldsymbol{\theta}^M]$, and the variance measuring the uncertainty in this approximation is given by the right-hand side of (3.3) with \mathbf{z}^* replaced by \mathbf{z} . Note that this variance is zero at the design points at which the function was actually evaluated.

The hyper-parameters $(\boldsymbol{\mu}^M, \boldsymbol{\theta}^M)$ are unknown, and will be dealt with in a Bayesian fashion. Because the posterior is no longer available in closed form, we have to generate a sample $(\boldsymbol{\mu}_j^M, \boldsymbol{\theta}_j^M)$ from this distribution using MCMC techniques. This will be detailed in Appendix B, with additional details available in Bayarri et al. (2005). The actual emulator of the computer model (and its variance) can be viewed as the average of (3.2) (and of (3.3)) over this posterior sample.

Key computational simplification: The major difficulty in the above computations is the inversion of the many thousands of matrices $\boldsymbol{\Gamma}_j^M$ that correspond to the covariance matrix $\boldsymbol{\Gamma}^M$ evaluated at each of the elements of the posterior sample; or, equivalently (and dropping the j subscripts), the inversions of the correlation matrices $\mathbf{C}^M(D^M \times D^T, D^M \times D^T)$. These matrices are of dimension $m = LN$ which, in CRASH, can be as large as 380. Additionally, the sampling mechanism utilized to obtain samples from the posterior distributions of the unknowns forces one to compute the inverse of the correlation matrix for values of the parameters that lead to highly ill-conditioned matrices. This leads to unstable calculations that become increasingly unreliable as the dimension of the matrices increases.

It is here that crucial use is made of the choice of a product form for the correlation function, together with the product input space $D^M \times D^T$. It follows that

$$\mathbf{C}^M(D^M \times D^T, D^M \times D^T) = \mathbf{C}_x^M(D^M, D^M) \otimes \mathbf{C}^T(D^T, D^T), \quad (3.4)$$

where \mathbf{C}_x^M and \mathbf{C}^T are correlation matrices corresponding to separate use of the \mathbf{x}

and t components of the correlation functions, and \otimes refers to the Kronecker product defined as: $\mathbf{A} \otimes \mathbf{B}$, for matrices $\mathbf{A}_{m \times n}$ and $\mathbf{B}_{p \times q}$, is the $mp \times nq$ matrix whose (i, j) block is $a_{ij}\mathbf{B}$, where a_{ij} is the (i, j) element of \mathbf{A} . The advantage of the Kronecker product structure (see Bernardo et al. (1992) for a related use) is that then

$$(\mathbf{C}^M(D^M \times D^T, D^M \times D^T))^{-1} = (\mathbf{C}_x^M(D^M, D^M))^{-1} \otimes (\mathbf{C}^T(D^T, D^T))^{-1}, \quad (3.5)$$

and inverting L and N -dimensional matrices (and multiplying the inverses together) is much cheaper, and more stable, than inverting an LN -dimensional matrix. Also, $|\mathbf{C}^M(D^M \times D^T, D^M \times D^T)| = |\mathbf{C}_x^M(D^M, D^M)|^N |\mathbf{C}^T(D^T, D^T)|^L$. These properties of the Kronecker product structure are also exploited in Williams et al. (2006) and Rougier (2007).

4 ANALYSIS OF MODEL OUTPUT

4.1 Notation and Statistical Modeling

Reality and bias: It is crucial to represent the computer model as a biased representation of reality, “reality = model + bias;” formally

$$y^R(\mathbf{x}, t) = y^M(\mathbf{x}, t) + b(\mathbf{x}, t), \quad (4.1)$$

where $y^R(\mathbf{x}, t)$ is the value of the ‘real’ process at input (\mathbf{x}, t) and $b(\mathbf{x}, t)$ is the unknown bias function.

Modeling the field response functions: The field response functions at input $\mathbf{x}^{*i}, i = 1, \dots, l$, are modeled as

$$y^F(\mathbf{x}^{*i}, t) = y^R(\mathbf{x}^{*i}, t) + \epsilon_i^F(t), \quad (4.2)$$

where the $\epsilon_i^F(t)$ are independent realizations from a Gaussian process prior with mean function 0, precision λ^F , and correlation function $c^T(t, t^*) = \exp\left(-\beta^T |t - t^*|^{\alpha^T}\right)$.

The assumption that $\epsilon^F(\cdot)$ has mean zero is formally the assumption that the field observations have no bias. The situation is otherwise quite problematic, in that there is then no purely data-based way to separate the field bias from the model bias. Estimates of bias that arise from our methodology could still be interpreted as the systematic difference between the computer model and field observations, but this is of little interest, in that prediction of reality (not possibly biased field data) is the primary goal. Assuming that each field response function error is itself a draw from a Gaussian process, and one of the same form (as a function of t) as the computer model, seems quite natural.

Stochastic modeling of the bias: Since we will be performing a Bayesian analysis, the bias must be assigned a prior distribution. It is natural to choose this to be another Gaussian process with constant mean function $\mu^b(\cdot) \equiv \mu^b$ and correlation function as in (3.1), but with its own set of hyper-parameters: $\boldsymbol{\alpha}^b$ and $\boldsymbol{\beta}^b$ being the correlation parameters associated with \boldsymbol{x} , and λ^b denoting the precision. (The correlation parameters associated with the t component are going to be discussed shortly.)

Since the bias cannot be directly observed, there is very little information available about the parameters governing the prior for the bias, and especially about the smoothness parameters. Also, numerical computations are more stable with these smoothness parameters fixed at pre-specified values. Empirically, in many examples we have looked at, CRASH being one of them, the maximum likelihood estimates of $\boldsymbol{\alpha}^b$ have mostly been near 2. Hence, we are going to restrict attention to the class of bias functions obtained by fixing all the components of $\boldsymbol{\alpha}^b$ at 2. Note that, as discussed below, the smoothness parameters associated with the t component are not

going to be fixed, which provides added flexibility in the prior for the bias.

The choice of a constant mean function for the Gaussian process of the bias was to allow some flexibility in the level of the bias, but to avoid confounding with the linear structure being assumed for the computer model Gaussian process. (Note that the sample posterior correlation between μ^M and μ^b is only -0.003 .)

Key assumption: We assume that the Gaussian process correlation parameters corresponding to the input t — namely α^T and β^T — are the same for the computer model, the field error, and the bias. As will be seen in Appendix A, this is a necessary assumption for computational implementation, in that it allows all the Bayesian computations to take advantage of the type of Kronecker product simplification illustrated in (3.5). Note that we are not assuming that the variations of the three functions with respect to t are the same; they are rather being assumed to have the same correlation structure.

In problems with very limited data, as in the CRASH example, the choices of the correlation parameters of the various processes are usually of secondary importance in the analysis; the precisions of the processes are much more influential. Nevertheless, to see if the assumption of constancy of α^T and β^T across the various processes is simply not tenable in the CRASH example, we separately determined their posterior distributions utilizing only the model-run data, utilizing only the field data, and utilizing all the data. The results are given in Figure 2.

There were only four effective runs available from the field data analysis (since only differences of replicates having exactly the same velocity inputs could be used), so close agreement between the various cases would not be expected. In this light, the agreement of the posteriors for β^T is quite satisfactory. The disagreement between the posteriors of α^T for model-run and field data is larger than one would like to see but, given the relative insensitivity of results to choices of α^T , proceeding

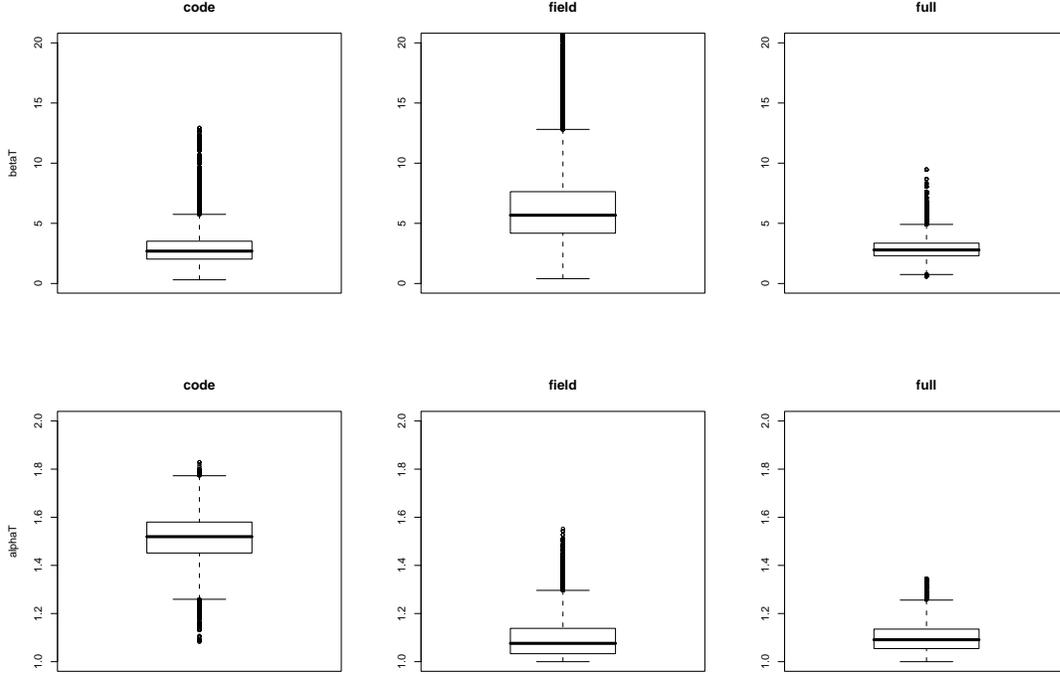


Figure 2: Box plots of posterior densities of β^T (top) and α^T (bottom) arising from utilizing only the model-run data (1st column), utilizing only the field data (2nd column), and utilizing all the data (3rd column).

with the assumption still seems to be a reasonable practical compromise, given the computational considerations.

4.2 Bayesian Analysis

To proceed with a Bayesian analysis, it is only necessary to specify a prior density for the remaining unknown parameters, namely $(\mu^M, \mu^b, \lambda^M, \lambda^b, \lambda^F, \alpha^T, \beta^T, \beta^M, \alpha^M, \beta^b)$, and apply Bayes theorem. The details of the prior assignment are given in Appendix A, and the Bayesian implementation via MCMC is discussed in Appendix B. Here, we focus on discussion of the possible outputs of the analysis.

The MCMC results in a sample of all unknowns, including the key functions $y^M(\mathbf{x}, t)$ and $b(\mathbf{x}, t)$ at specific input values \mathbf{x} of interest. (Note that equation (4.1) can then be used to produce a sample from $y^R(\mathbf{x}, t)$ as well.) Denote the MCMC

sample by $\{y_j^M(\mathbf{x}, t), b_j(\mathbf{x}, t) : j = 1, \dots, S\}$. A technical point is that one must again discretize t and obtain the function predictions at these discretized values; but this can be done at a much finer set of t values than D^T , since the computations involved in prediction are much faster than those involved in obtaining a posterior sample of unknown parameters. In CRASH, the discretization $D_t^P = \{3, 6, \dots, 81\}$ was used at this prediction stage, and was quite adequate for reconstruction of the functions. Details of this aspect of the calculations are in Appendix B.

A final point is that, when predicting curves for initial velocities that are not part of the model and field data that were originally collected, we must introduce the added information that the initial relative velocity is zero (else the posterior realizations would, inappropriately, have varying initial relative velocities). Incorporation of such constraints is straightforward, as explained in detail in Appendix B.

4.2.1 Bias estimates

The estimated bias function is obtained from the posterior sample as $\hat{b}(\mathbf{x}, t) = S^{-1} \sum_{j=1}^S b_j(\mathbf{x}, t)$. This estimated bias function for SDM velocity at the input $x_1 = 56.3 \text{ km/h}$ (impact velocity) is the central curve in Figure 3.

It is also important to give tolerance or confidence bands for any estimated functions. Pointwise 80% posterior intervals for the bias are also given in Figure 3. These were found by simply taking the 10th and 90th percentiles of the S posterior bias sample functions at each (discretized) t . Note that these confidence bands are wide enough that there is no clear indication of bias at the given input value.

In contrast, Figure 4 shows the posterior estimate of the bias for a 30 km/h impact, along with the 80% confidence bands. The bias is clearly larger in the $20\text{-}59 \text{ ms}$ interval than it was for the 56.3 km/h impact, and now the confidence bands do not cover 0. Hence this is clear evidence that the computer model is biased at these input values. This was also noticed in other contexts, leading to the conclusion that the computer

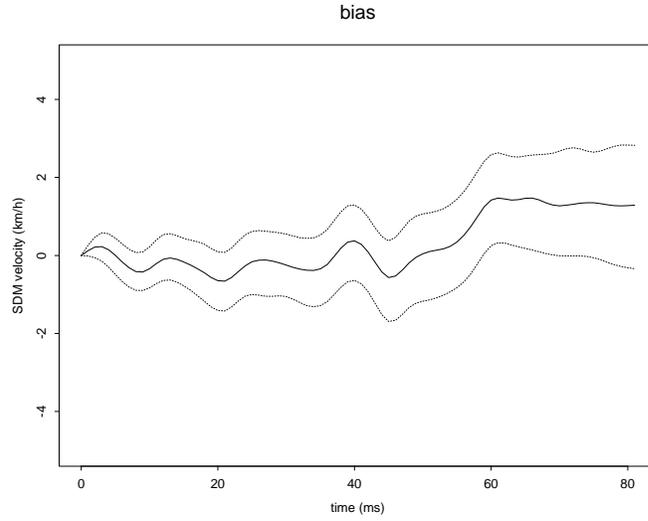


Figure 3: Estimate and 80% posterior intervals for SDM velocity bias, at 56.3km/h .

model appears to be more severely biased at low impact velocities than at higher impact velocities, a finding of considerable interest to the modelers.

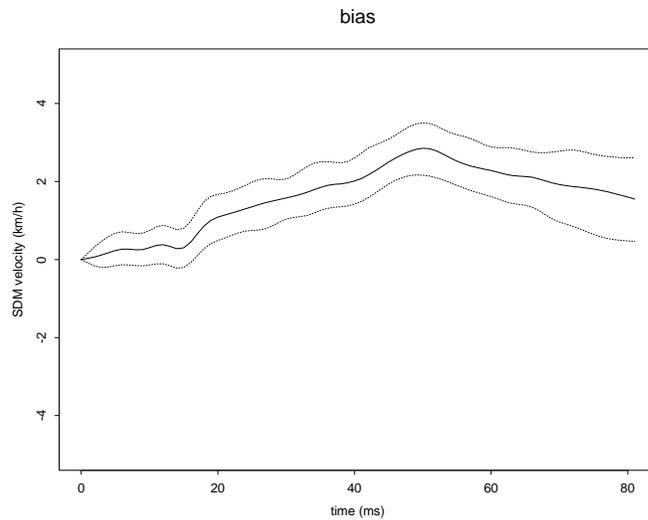


Figure 4: Estimate and 80% posterior intervals for SDM velocity bias, at 30km/h .

4.3 Predicting the Computer Model and Reality

Estimates of $y^M(\mathbf{x}, t)$ and $y^R(\mathbf{x}, t)$ at a (new) input \mathbf{x} are given by the MCMC estimate of the posterior means of the functions, namely

$$\hat{y}^M(\mathbf{x}, t) = \frac{1}{S} \sum_{j=1}^S y_j^M(\mathbf{x}, t) \quad \text{and} \quad \hat{y}^R(\mathbf{x}, t) = \frac{1}{S} \sum_{j=1}^S y_j^R(\mathbf{x}, t), \quad (4.3)$$

where $y_j^R(\mathbf{x}, t) = y_j^M(\mathbf{x}, t) + b_j(\mathbf{x}, t)$ is a posterior sample of the real process. The estimate $\hat{y}^R(\mathbf{x}, t)$ is called the *bias-corrected prediction of reality* in Bayarri et al. (2007b) because it equals $\hat{y}^M(\mathbf{x}, t) + \hat{b}(\mathbf{x}, t)$.

This analysis assumes that the computer model is not run at the new input \mathbf{x} ; the situation when it is subsequently run at \mathbf{x} (resulting in a new data point for the analysis) is considered in Appendix B.

Posterior confidence bands for both estimates can also be computed, using the samples generated from the MCMC. The estimates and confidence bands are given in Figure 5 for SDM velocity when the impact is 56.3km/h .

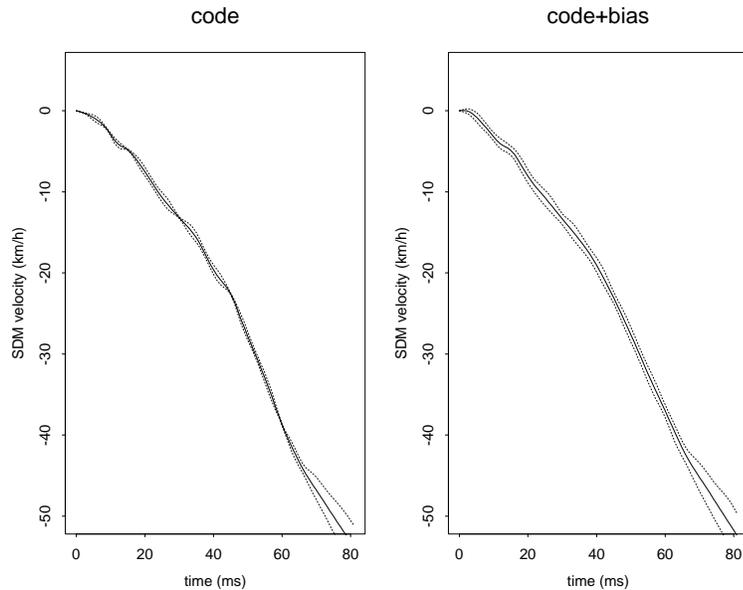


Figure 5: Estimate and 80% posterior intervals of SDM velocity for y^M (left figure) and for y^R (right figure), when the impact velocity is 56.3km/h .

The tight confidence bands around \hat{y}^M indicate that the GASP approximation to the computer model code is quite accurate at the $56.3km/h$ input, except for times exceeding 65ms. The bias-corrected prediction of reality also appears to be quite accurate over the same region.

The confidence bands for the computer model prediction are not indicators of the accuracy of the computer model in predicting reality; they simply indicate the accuracy of the GASP approximation to the computer model. Overall accuracy of the computer model prediction, $\hat{y}^M(\mathbf{x}, t)$, can be inferred indirectly from the posterior distribution of the bias. We have found it useful in communication with engineers, however, to also quantify this accuracy by producing *tolerance bands*, as in Bayarri et al. (2007b). Tolerance bands are constructed so that one can, e.g., make the statement “with probability 0.80, the prediction $\hat{y}^M(\mathbf{x}, t)$ is within a specified tolerance τ of the true $y^R(\mathbf{x}, t)$.” (Note that this is a pointwise statement.) Symmetric 80% tolerance bands are simply found by choosing τ so that 80% of the posterior samples satisfy $|\hat{y}^M(\mathbf{x}, t) - [y_j^M(\mathbf{x}, t) + b_j(\mathbf{x}, t)]| < \tau$.

4.4 Prediction of CRITV

Recall that one important evaluation function for CRASH was CRITV, the SDM velocity calculated 30ms before the time the SDM displacement (DISP) reaches 125mm. Note that DISP is just integrated velocity, i.e., $DISP(t) = -\int_0^t y^R(\mathbf{x}, v)dv$. Thus $CRITV = y^R(\mathbf{x}, DISP^{-1}(125) - 30)$. (We will suppress the dependence of CRITV on \mathbf{x} in the notation.)

Obtaining a posterior sample for CRITV is relatively straightforward. For bias-corrected prediction of reality, one takes each sample function $y_j^R(\mathbf{x}, t)$ from the posterior and simply solves for the corresponding $CRITV_j$ using the above formulas. The result is a sample $\{CRITV_j : j = 1, \dots, S\}$ from the posterior distribution of CRITV. (Note that CRITV is a very involved function of the other parameters, yet

the MCMC technique produces its posterior very easily.) Likewise, one could use the posterior sample $\{y_j^M(\mathbf{x}, t) : j = 1, \dots, S\}$ of predictions of the computer model to estimate CRITV^M , by which we mean the value of CRITV that would result from actually exercising the computer model at \mathbf{x} and computing CRITV from the resulting $y^M(\mathbf{x}, t)$.

As an illustration, Figure 6 gives the posterior distribution of CRITV (lower figure) and CRITV^M (upper figure) at a 56.3km/h impact velocity.

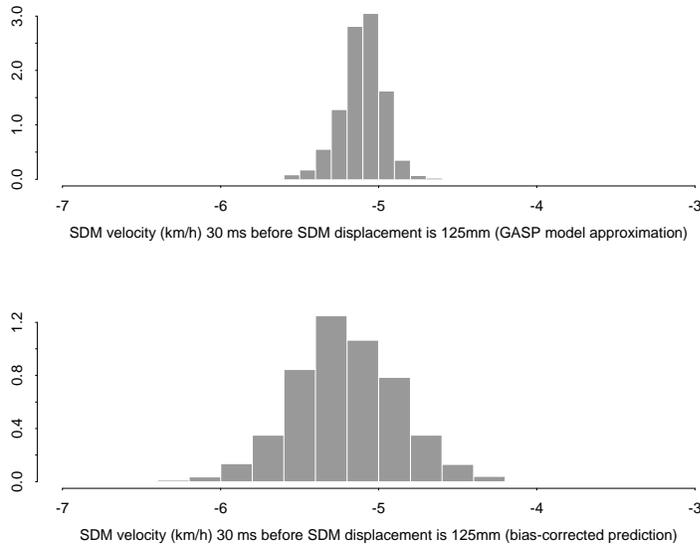


Figure 6: Posterior distribution of CRITV (lower figure) and CRITV^M (upper figure), at a 56.3km/h impact velocity.

The fact that the posterior for CRITV^M is quite tight shows that using the GASP approximation to the computer model is effective. Of course, the posterior for CRITV^M does not inform directly about the quantity of interest, CRITV, and the lower figure shows that there is considerably more uncertainty as to its actual value. Indeed, 80% tolerance bands for CRITV^M are -5.11 ± 0.17 , while the 80% bias-corrected prediction bands corresponding to the lower figure are the much broader -5.21 ± 0.42 .

5 HIERARCHICAL MODELING FOR RELATED SITUATIONS

In trying to extrapolate prediction to a new scenario in which only limited data is available, one can use hierarchical modeling. Hierarchical modeling applies most directly to scenarios in which there are K different function outputs, to be denoted by subscripts i , each coming from different configurations of a computer model (or even from different computer models); in the CRASH example, these differing configurations are the differing impact angles and barrier types. Related analyses can be found in Han et al. (2009) and Higdon and Gattiker (2008). An alternative approach, using multivariate Gaussian processes to directly model the multivariate outputs, can be found in Qian et al. (2008) and in Conti and O’Hagan (2007).

Each of the different model configuration output functions can be modeled as was done in sections 3 and 4, through Gaussian process priors. We will be particularly concerned with settings where the Gaussian processes for y_i^M and b_i can be assumed to share common features, typically, where the parameters governing the priors are drawn from a common distribution. This induces connections among the individual models and enables us to combine information from the separate models, sharpen analyses, and reduce uncertainties.

Implementation of these ideas will depend heavily on what data, both computer and field, are available as well as the legitimacy of the assumptions imposed. Here, we state and comment on these assumptions for the simplest structure we will impose, leaving some of the details to Appendix C.

Assumption 1. The correlation parameters of the model approximation processes are identical across the K configurations being considered; i.e., all computer model and field data have common α ’s and common β ’s, which are assigned priors as in

the single-model case. This assumption was made because the CRASH data is too limited to allow separate determination of the correlation parameters for each model configuration, and the engineers involved in the study viewed the assumption as reasonable. (The data is also too limited to provide any evidence that this assumption might be incorrect.) In other contexts, this assumption may well be inadvisable.

Assumption 2. The variances $1/\lambda^M$ of the model approximation processes are equal, across the various cases, as are the variances $1/\lambda^F$ of the field data. The priors used for these parameters are as in the single-model case. If one had sufficient data, relaxing this assumption (i.e., allowing differing variances following a hierarchical model) would be natural (and likely more important than relaxing Assumption 1).

Assumption 3. The unknown mean parameters, μ_i^M , of the GASP approximations for the K computer model configurations, are assumed to arise from the two-stage hierarchical model detailed in Appendix C. This prior distribution will allow the computer model outputs to vary considerably between the K configurations, but still ensures that information is appropriately pooled in their estimation.

Assumption 4. The bias process means, μ^b , are less important in the analysis than the bias process precisions, λ_i^b , so having the latter vary between model configurations was felt to be allowing sufficient variation in the bias processes, given the limited amount of available data. The bias process means for the K configurations are set equal to a common (unknown) parameter μ^b with a prior as in the single-model case.

The variances of the bias processes are related in a fashion described by a parameter q , whose value must be specified. Specifically, we assume that $\log(\lambda_i^b) \sim N(\eta, 4q^2)$, with specified q and a constant prior assigned to η . Parameter q describes the believed degree of similarity in the biases for the K different computer model configurations; indeed, $1 + q$ can be interpreted as an upper bound on the believed ratio of the stan-

dard deviations of the biases, or, stated another way, the proportional variation in the bias is q . For instance, $q = 0.1$ implies that the biases are expected to vary by about 10% among the various cases being considered.

Note that specification of q is a judgment as to the *comparative* accuracy of the K different computer model configurations, as opposed to their *absolute* accuracy (which need not be specified). The reason we require specification of q by the engineer/scientist is that there is typically very little information about this parameter in the data (unless K is large). Specifying q to be zero could be reasonable, if one is unsure as to the accuracy of the computer models but is quite sure that the accuracies are the same across the various K .

Finally, the correlation parameters of the bias processes are assumed to be common across the K different configurations, and are assigned priors as in the single-model case; in particular, the smoothness parameters are set equal to 2.

The analysis reported in Section 4 was for the data and model for rigid barrier, straight frontal impact. By use of hierarchical modeling we can simultaneously treat rigid barrier, left angle and right angle impacts as well as center pole impact. Thus we use the hierarchical model with $K = 4$ related situations. The analyses and predictions reported below are for a $56.3\text{km}/h$ impact (this is at the high end of the data). The hierarchical model was used with $q = 0.1$. For simplicity, however, the prior distributions used for the GASP parameters were chosen to be the same as those used for the straight frontal analysis (and described in Appendix A); this is reasonable, since the priors are relatively non-informative and the straight frontal dataset is by far the largest of the four categories.

Figure 7 shows the posterior distributions of $\log(\lambda_i^b)$ and μ_i^M for individual barrier types. Note that, while the assumed similarity between the models allows information to be passed from ‘large data’ to ‘small data’ models, the models are still allowed to vary significantly.

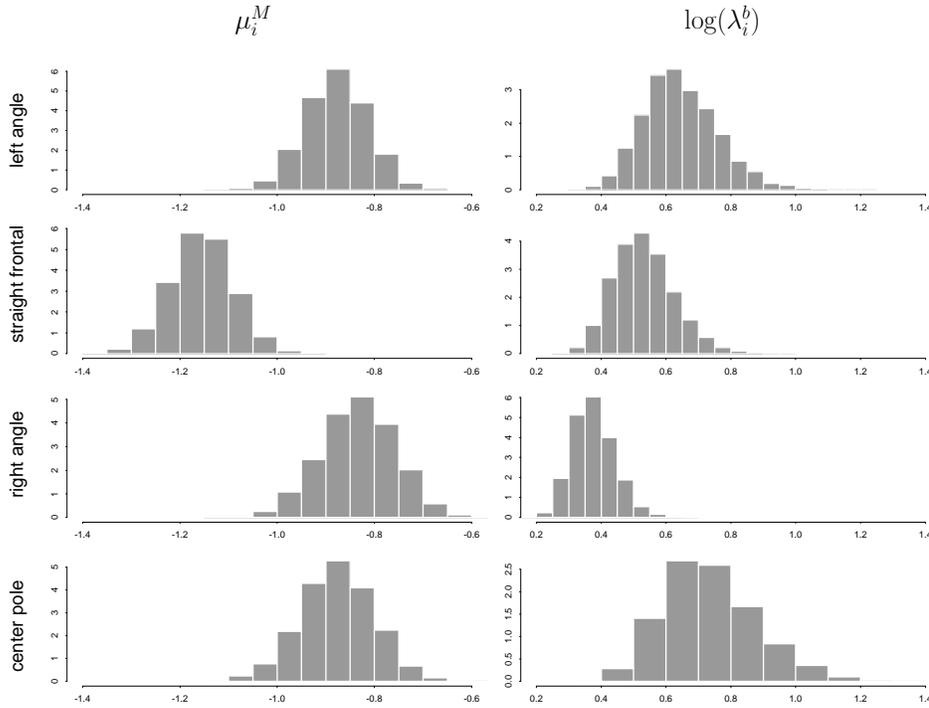


Figure 7: Posterior distributions of μ_i^M and $\log(\lambda_i^b)$ for the 4 barrier types, based on the hierarchical model.

Figure 8 shows the differing posterior predictive SDM velocity curves and point-wise uncertainty bands for each of the four barrier types. The straight frontal and left angle posterior intervals in Figure 8 are tight, in part because there are data with inputs close to 56.3km/h for these barrier types. In contrast, the intervals are not tight for the other two barrier types because data near 56.3km/h are lacking. (This thus reinforces the potential value of making a model run at a new desired input.) Note that Figure 5 and the straight frontal pictures in Figure 8 are similar, so that the hierarchical analysis did not greatly affect the answers for this barrier type.

Figure 9 gives the estimates of the four bias functions and the associated pointwise uncertainties. Because of the large uncertainties in the bias estimates, the only case in which the bias seems clearly different from zero is for left angle impacts, after 43ms . (While we cannot clearly assert that there is bias in the other cases, the

tolerance bounds for predictions will be quite large, reflecting the uncertainty in the bias estimates.)

For CRITV, Table 2 presents, for each barrier type, the posterior mean and standard deviations of $CRITV^M$ and CRITV arising under the GASP model approximation and the bias-corrected prediction, respectively.

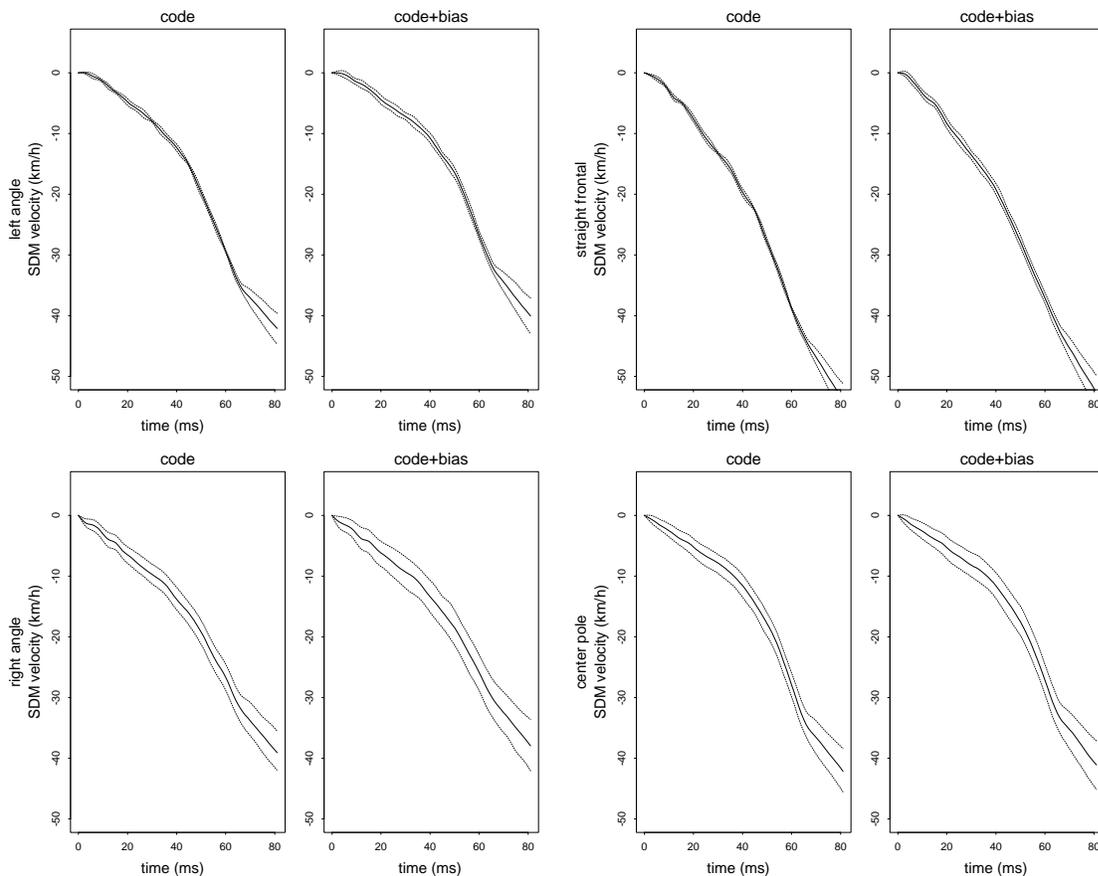


Figure 8: Pointwise 80% posterior intervals for SDM velocity under 4 barrier types, based on the hierarchical model, and arising from the GASP model approximation estimate and the bias-corrected prediction, respectively, at the input velocity $56.3\text{km}/h$.

If we only consider the three rigid barrier impacts (frontal, right angle and left angle) and ignore the center pole impact data, we could proceed without use of hierarchical modeling by incorporating the angle of impact, x_2 , as an input to the model. The smoothness assumption required for the Gaussian process analysis is plausible: it is reasonable to assume that small changes in the angle will result in

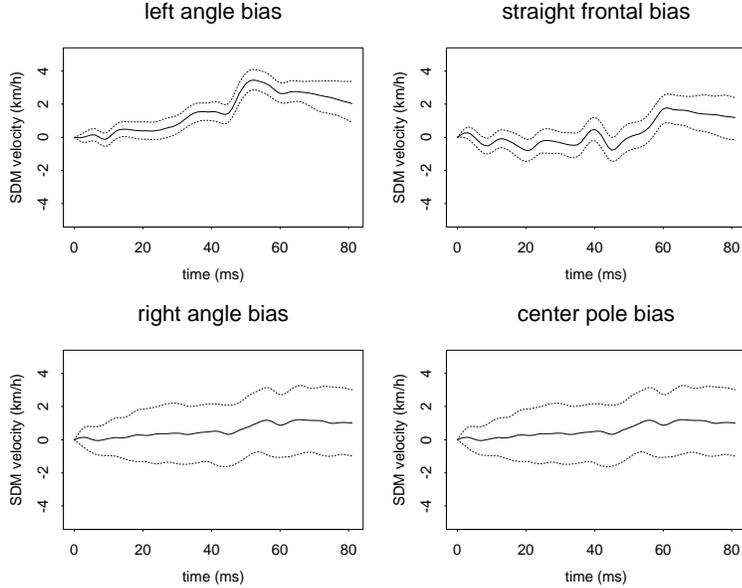


Figure 9: Estimates and pointwise 80% posterior intervals for bias, under 4 barrier types, based on the hierarchical model and at the input velocity $56.3\text{km}/h$.

Barrier type	Hierarchical model		Using frontal data only	
	CRITV ^M	CRITV	CRITV ^M	CRITV
left angle	-6.08 (0.34)	-6.34 (0.49)		
straight frontal	-5.13 (0.13)	-5.22 (0.30)	-5.11 (0.13)	-5.21 (0.33)
right angle	-6.89 (0.65)	-6.80 (0.96)		
center pole	-6.55 (0.74)	-6.54 (0.91)		

Table 2: Posterior mean and standard deviation of CRITV^M and CRITV, at the input velocity $56.3\text{km}/h$.

small changes in the velocity-time curve so that y^M is a smooth function of x_2 . We performed this analysis with the input vector now being $\mathbf{x} = (x_1, x_2)$, but the results differed little from the results in the hierarchical analysis and so are not reported. The computation was also considerably more intensive via this route.

6 FORMAL TESTING OF MODEL VALIDITY

A question often asked in the computer model validation community is “Is the computer model correct?” While an increasing segment of the community understands

that the answer is ‘no,’ it is of considerable pedagogical interest to provide methodology for clear demonstration of the answer. Bayesian testing (in which bias is allowed) seems to be extremely powerful in addressing this question formally, and the goal of this section is to present the relevant Bayesian testing methodology.

Letting \mathcal{M}_0 denote the computer model, a natural way to approach the question is to formally test the hypothesis $H : “\mathcal{M}_0 \text{ is true}”$. To test this hypothesis within the Bayesian framework, an alternative model is needed. Luckily, we have a ready-made alternative, \mathcal{M}_1 , namely the model constructed in sections 3 and 4, which includes the bias term $b(\mathbf{x}, t)$. To perform the test, we thus need only compute the posterior probability that \mathcal{M}_0 is correct.

Let ϕ_i be the entire parameter vector for model \mathcal{M}_i , $i = 0, 1$ (including all parameters of the mean functions and the Gaussian processes involved). In addition, for model \mathcal{M}_i , $i = 0, 1$, denote by $f_i(\mathbf{y} | \phi_i)$, $p_i(\phi_i)$, and $p_i(\phi_i | \mathbf{y})$ the likelihood function of the full data vector \mathbf{y} (both computer model and field data), the prior density, and the posterior density for the parameter vector, respectively. The form of the likelihood function and the approaches for prior specification and posterior inference, using MCMC methods, for model \mathcal{M}_0 are similar to the corresponding ones for model \mathcal{M}_1 , described earlier and detailed in appendices A and B, with one exception. Use of improper priors is typically not possible when interest lies in computation of the posterior probability of models. This is a potential concern because the Bayesian predictive analysis discussed previously assigned constant improper priors to both μ^M and μ^b . However, the parameter μ^M is a location-parameter that occurs in both models and so can be assigned the constant prior, as justified in Berger et al. (2001).

Unfortunately, μ^b only occurs in model \mathcal{M}_1 , and so it cannot be assigned a constant prior. Hence, for the computation in this section, we simply assumed that $\mu^b = 0$.

Letting $\pi_1 = 1 - \pi_0$ denote the prior probability of model \mathcal{M}_1 , Bayes theo-

rem yields that the posterior probability of model \mathcal{M}_0 is given by $P(\mathcal{M}_0 \mid \mathbf{y}) = \pi_0 m_0(\mathbf{y}) / [\pi_0 m_0(\mathbf{y}) + \pi_1 m_1(\mathbf{y})]$, where

$$m_i(\mathbf{y}) = \int f_i(\mathbf{y} \mid \boldsymbol{\phi}_i) p_i(\boldsymbol{\phi}_i) d\boldsymbol{\phi}_i \quad (6.1)$$

is the marginal likelihood for model \mathcal{M}_i , $i = 0, 1$.

Although we can analytically integrate in (6.1) over part of the parameter vector $\boldsymbol{\phi}_i$, part of the integration must be done numerically. There are a variety of possible methods for such numerical integration (see, e.g., Han and Carlin (2001)). We considered two approaches: that of Chib and Jeliazkov (2001); and importance sampling, utilizing a t distribution (with mean and variance estimated from the MCMC output) as the importance function (see Bayarri et al. (2005) for the details).

For $\pi_0 = 1/2$, the two computational approaches lead to posterior probabilities for model \mathcal{M}_0 of 2.694×10^{-26} and 1.334×10^{-23} , respectively. The computation via importance sampling was stable and accurate (which is easy to assess because the importance samples are i.i.d.), but the computation using the method of Chib and Jeliazkov did not seem to fully stabilize; the convergence process appears to have very heavy tails.

Finally, it should be mentioned that it is often more reasonable to approach formal testing of validity by simply testing if the bias is suitably small, say, $|b(\mathbf{x}, t)| < \delta$ for all t in an interval of interest, and for some specified δ . This could easily be done using the posterior samples of the bias, as in Section 4.2.1.

7 SUMMARY

7.1 Conclusions for CRASH and Engineering

The basic question of interest to the engineers in the study was whether CRASH reliably reproduces the results of physical tests of vehicle crashworthiness. It was found that CRASH is reasonably reliable (with quantified uncertainty) for larger impact velocities, but has a definite bias at lower impact velocities. This was of considerable interest and initiated a re-examination of the computer model to determine the source of this bias. The development of the bias-corrected prediction methodology for CRASH also allowed use of the current CRASH model, in spite of the bias.

The quantity CRITV was of particular importance for the users of the code, as it related to air bag firing times. Using the bias-corrected prediction methodology, it was not only possible to estimate CRITV, but also to assess the uncertainty in the estimate (crucial for a design issue as sensitive as air bag firing); such uncertainties were not previously available.

The emulator of the computer model that was constructed, as part of the analysis, was also found to be very useful for exploring the design space in a much more efficient way than could the CRASH model itself (due to the prohibitive run-times of CRASH).

Finally, this case-study was an important step in the development of a general set of tools that are now being used by engineers in the validation of a variety of computer models constructed for vehicle design.

7.2 Summary of Technical Innovations

The methodological extensions of the framework described in Bayarri et al. (2007b), to meet the challenges imposed by the problem of validating the CRASH model, are potentially useful for other applications that involve (time-dependent) functional data that are suitably smooth. By discretizing the model output and field curves at

a relatively small number of time points, it is possible to incorporate functional data by the direct method of treating time as an additional input to the model and field processes. To overcome the computational issues that could result from the explosion of input values, a strategy based on a Kronecker product formulation of the covariance matrices was introduced. To be computationally successful, the strategy requires that the various Gaussian processes used in the analysis have a common correlation structure in terms of time. (While this assumption is plausible for the CRASH model, it is certainly an assumption to be considered carefully in other applications.)

A further methodological contribution was the way of incorporating constraints – such as the restriction that relative velocities are zero at time zero – into the analysis.

The CRASH dataset comprises limited data regarding impact configurations other than the straight frontal. Another technical innovation was thus to construct a hierarchical model that is able to appropriately borrow strength across the different configurations. Strong assumptions were required but, in such data-limited situations, there is no real alternative.

Finally, a method was developed for formally testing the hypothesis that the computer model is an adequate representation of reality, although we argued that such tests are of limited utility.

SUPPLEMENTAL MATERIALS

Technical Report: “NISS-tech-report-163.pdf” is the Bayarri et al. (2005) National Institute of Statistical Sciences technical report, referenced in Sections 1.5 and 6, and Appendixes B and C. (pdf file)

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A The Model

The Likelihood: Let $\mu^M(\mathbf{z}) = \Psi(\mathbf{z})'\boldsymbol{\mu}^M$ be the mean of the Gaussian process prior for the computer model, where $\Psi(\mathbf{z})$ is an r -dimensional known function of \mathbf{z} . This general formulation includes as a special case the linear trend form used for the CRASH model (and given in Section 3).

Given the unknown parameters, the data $\mathbf{y} = (\mathbf{y}^{M'}, \mathbf{y}^{F'})'$ is normally distributed with mean vector $\mathbf{X}\boldsymbol{\theta}$ and covariance matrix $\boldsymbol{\Sigma} \otimes \mathbf{C}^T(D^T, D^T)$ where $\boldsymbol{\theta} = ((\boldsymbol{\mu}^M)', \boldsymbol{\mu}^b)'$,

$$\mathbf{X} = \begin{pmatrix} \Psi(\mathbf{z}_1)' & 0 \\ \vdots & \vdots \\ \Psi(\mathbf{z}_m)' & 0 \\ \Psi(\mathbf{z}_1^*)' & 1 \\ \vdots & \vdots \\ \Psi(\mathbf{z}_n^*)' & 1 \end{pmatrix}, \quad (\text{A-1})$$

$$\boldsymbol{\Sigma} = \begin{pmatrix} \frac{1}{\lambda^M} \mathbf{C}^M(D^M, D^M) & \frac{1}{\lambda^M} \mathbf{C}^M(D^M, D^F) \\ \frac{1}{\lambda^M} \mathbf{C}^M(D^M, D^F)' & \frac{1}{\lambda^M} \mathbf{C}^M(D^F, D^F) + \frac{1}{\lambda^b} \mathbf{C}^b(D^F, D^F) + \frac{1}{\lambda^F} \mathbf{I}_{n \times n} \end{pmatrix}. \quad (\text{A-2})$$

Prior distributions: Since $\boldsymbol{\theta}$ is a location vector, we utilize the standard constant prior density $p(\boldsymbol{\theta}) = 1$. It thus remains only to specify the prior density for $\boldsymbol{\xi} = (\lambda^M, \lambda^b, \lambda^F, \boldsymbol{\beta}^M, \boldsymbol{\beta}^b, \boldsymbol{\beta}^T, \boldsymbol{\alpha}^M, \boldsymbol{\alpha}^T)$.

For the smoothness parameters α^M and α^T , we choose uniform priors on their ranges (each α_ℓ being in the interval $(1, 2)$). The precision parameters $(\lambda^M, \lambda^b, \lambda^F)$ and range parameters $(\beta_\ell^M, \beta_\ell^b, \beta^T)$ are given independent exponential priors, with means set at 10 times the marginal likelihood estimate of the parameter. This was essentially an empirically driven choice. When the mean was set at 10 times the marginal MLE, the prior had little influence on the answer (so that the ‘double use’ of the data in defining the centering of the prior is not a significant sin), and the exponential tail did help considerably with the convergence of the MCMC.

The marginal MLE of ξ is defined as the MLE from the integrated likelihood, obtained by integrating out θ with respect to its prior. Explicit formulas for this integrated likelihood, for the associated Fisher information matrix, and details about the utilization of Fisher’s scoring method in computing the estimate can be found in Berger et al. (2001) and Paulo (2005).

In the CRASH application, the marginal MLE of $(\lambda^M, \lambda^b, \lambda^F, \beta^T, \beta^M, \beta^b)$ is $(0.051, 0.77, 1.24, 2.63, 0.25, 31.7)$, and hence the prior on, e.g., β^T is an exponential distribution with mean 10×2.63 . The marginal MLEs for α^T and α^M are, respectively, 1.10 and 1.03 (recall that α^b was fixed at 2), but these are not used in the prior specification since these parameters are *a priori* uniform on their ranges.

B Monte Carlo methods for posterior inference

Generating samples from the posterior distribution of the unknown parameters:

The form of covariance matrix $\Sigma \otimes C^T(D^T, D^T)$ does not result in simple expressions for full conditional distributions of parameters contained in ξ , so we utilized a block Gibbs sampler to generate from the posterior distribution. We work with two full conditionals, $p(\theta \mid \xi, \mathbf{y})$ and $p(\xi \mid \theta, \mathbf{y})$, and, given the current draw $(\theta^{(\text{old})}, \xi^{(\text{old})})$, update according to (a) $\theta^{(\text{new})} \sim p(\theta \mid \xi^{(\text{old})}, \mathbf{y})$; and (b) $\xi^{(\text{new})} \sim p(\xi \mid \theta^{(\text{new})}, \mathbf{y})$. Step

(a) is easily implemented since $p(\boldsymbol{\theta} \mid \boldsymbol{\xi}, \mathbf{y})$ is a multivariate normal density with mean vector $(\mathbf{X}'(\boldsymbol{\Sigma} \otimes \mathbf{C}^T(D^T, D^T))^{-1} \mathbf{X})^{-1} \mathbf{X}'(\boldsymbol{\Sigma} \otimes \mathbf{C}^T(D^T, D^T))^{-1} \mathbf{y}$ and covariance matrix $(\mathbf{X}'(\boldsymbol{\Sigma} \otimes \mathbf{C}^T(D^T, D^T))^{-1} \mathbf{X})^{-1}$.

To sample from $p(\boldsymbol{\xi} \mid \boldsymbol{\theta}, \mathbf{y})$, we use a Metropolis-Hastings algorithm (e.g., Robert and Casella (2004)). To facilitate the choice of the proposal distribution, we transform $\boldsymbol{\xi}$ to $\boldsymbol{\xi}^* = g(\boldsymbol{\xi})$, where

$$g(\boldsymbol{\xi}) = \left(\ln[(\lambda^M, \lambda^b, \lambda^F, \boldsymbol{\beta}^{M'}, \boldsymbol{\beta}^{b'}, \beta^T)'], \ln \left(\frac{\boldsymbol{\alpha}^M - 1}{2 - \boldsymbol{\alpha}^M} \right), \ln \left(\frac{\alpha^T - 1}{2 - \alpha^T} \right) \right)$$

and the operations involving vectors should be interpreted componentwise.

A proposal distribution with which we have had good empirical results arises from a t -density with ν degrees of freedom, centered at $g(\boldsymbol{\xi}^{(\text{old})})$, and with scale matrix proportional to $I^{-1}(\hat{\boldsymbol{\xi}}^*)$. Here, $\hat{\boldsymbol{\xi}}^*$ is the (marginal) MLE for $\boldsymbol{\xi}^*$, obtained from the (marginal) MLE $\hat{\boldsymbol{\xi}}$ for $\boldsymbol{\xi}$, and $I(\hat{\boldsymbol{\xi}}^*)$ is the estimated information matrix, obtained from $I(\hat{\boldsymbol{\xi}})$, the corresponding matrix for $\boldsymbol{\xi}$ (both discussed in Appendix A). Note that the actual proposal distribution for $\boldsymbol{\xi}$ is induced by the multivariate t proposal for $\boldsymbol{\xi}^*$, through the transformation g , and is not symmetric. To choose the constant c to multiply $I^{-1}(\hat{\boldsymbol{\xi}}^*)$, we can start by trying $c = 2.4/\sqrt{d}$, where d is the dimension of $\boldsymbol{\xi}$, as suggested by Gelman et al. (1995), and then increasing or decreasing c in order to achieve a suitable acceptance rate. For CRASH, good mixing was achieved for $\nu = 5$ and $c = 1.5$.

Simulating realizations from the posterior distribution of functions: One usually wants to predict functions such as $y^M(\mathbf{x}, t)$, $y^R(\mathbf{x}, t)$ and $b(\mathbf{x}, t)$ at a finer set of t values than D^T . This is possible because the computations involved in prediction are much faster than those involved in obtaining a posterior sample of unknown parameters. Thus suppose prediction is desired at the n_P time points $D_t^P = \{t_1, \dots, t_{n_P}\}$. (Note that Kronecker product simplifications will not then be possible, but that is

not a problem here.) If prediction is desired at input \mathbf{x} , it follows that we need to predict the functions at the combined inputs $D^P = \{(\mathbf{x}, t_1), \dots, (\mathbf{x}, t_{n_p})\}$.

Hence, we need to generate a sample from the posterior predictive distribution of the combined vector $\mathbf{r}(D^P) = (\mathbf{y}^M(D^P)', \mathbf{b}(D^P)')'$, where $\mathbf{y}^M(D^P)$ and $\mathbf{b}(D^P)$ denote the vectors of sampled y^M and b function points respectively. Conditional on $(\boldsymbol{\theta}, \boldsymbol{\xi})$, the joint distribution of $\mathbf{r}(D^P)$ and the data is multivariate normal with readily computed mean and covariance vectors. The conditional distribution for $\mathbf{r}(D^P)$ is therefore easy to obtain as well. Hence, to draw a sample from the desired posterior predictive distribution, for every MCMC sample $(\boldsymbol{\theta}_j, \boldsymbol{\xi}_j)$, we only need to draw from the ensuing conditional multivariate normal.

The results reported for CRASH were based on MCMC posterior samples obtained by discarding the first 10,000 iterations (burn-in), storing the next 40,000 sampled values, and selecting every 40th point from this collection. The resulting 1,000 posterior samples for the parameters were then used to generate 1,000 realizations of $y^M(\mathbf{x}, t)$ and $b(\mathbf{x}, t)$ at the inputs \mathbf{x} of interest, for t in $D_t^P = \{3, 6, \dots, 81\}$.

Updating: The model approximation is exact only at the observed model-run data points. Sometimes the values of the model output are also constrained at other points. For instance, in CRASH, all relative velocity curves are (by definition) equal to zero at $t = 0$. If one tried to incorporate this constraint *a priori*, the Kronecker product computational simplification would no longer apply, resulting in an impractical MCMC algorithm. Another crucial instance of the conditioning idea is when an additional computer model run, $y^M(\mathbf{x}, t)$, becomes available. Indeed, this is how the computer model is often utilized for a new input \mathbf{x} of interest. The difficulty here is that it may not be feasible to re-run the entire MCMC algorithm with this new data point.

The solution to both these problems is simply to condition the existing poste-

rior on the additional constraint or data point; i.e., use the additional information in the Kalman filter part of the analysis, but not to obtain the posterior for tuning parameters or parameters in the Gaussian processes. To illustrate this approach, consider updating the posterior by the constraint that the relative velocity is 0 at $t = 0$. This is equivalent to introducing the ‘new (extended) observations’ $y^M(x_1, 0) = y^R(x_1, 0) = 0$. Let Σ^+ denote the prior covariance matrix of the new data vector $\mathbf{y}^+ = (\mathbf{y}', y^M(x_1, 0), y^R(x_1, 0))'$. The Σ^+ can be written in a block format to take advantage of the already computed inverses for Σ and $\mathbf{C}^T(D^T, D^T)$, and of the Kronecker structure. Posterior quantities can then be recalculated by replacing $(\Sigma \otimes \mathbf{C}^T(D^T, D^T))^{-1}$ by $(\Sigma^+)^{-1}$ and replacing \mathbf{y} by \mathbf{y}^+ . More details can be found in Bayarri et al. (2005).

C Hierarchical model assumptions and analysis

Suppose we have K related models. For the i th model, we use the subscript i for the parameters μ^M , and λ^b that are allowed to vary between models. The assumptions made in the hierarchical analysis are as follows.

1. All models and field data have common α 's, common β 's, common μ^b 's, common λ^M 's, and common λ^F 's, with priors as in Appendix A.
2. $\mu_i^M \mid \mu, \Upsilon \sim N_r(\mu, \Upsilon)$. (Recall that r is the dimension of the linear model for the mean of the Gaussian process prior for the model output.) The hyperparameters μ and Υ are assigned a reference prior, written in terms of μ and $\Omega = \lambda^M \Upsilon$, following Yang and Chen (1995). When $r = 1$ as for CRASH, so that the hierarchical prior is $\mu_i^M \mid \mu, \tau \sim N(\mu, \tau^{-1})$, this reference prior can be shown to be

$$p(\mu, \tau \mid \lambda^M, \alpha^M, \beta^M, \alpha^T, \beta^T) = \frac{1}{\tau} \left| \sum_{i=1}^K \left(1 + \frac{\tau}{\lambda^M \mathbf{X}_i^M \mathbf{C}_i^{-1} \mathbf{X}_i^{M'}} \right)^{-2} \right|^{1/2},$$

where $\mathbf{X}_i^M = (\Psi(\mathbf{z}_{i1})', \dots, \Psi(\mathbf{z}_{im_i})')'$, and $\mathbf{C}_i = \mathbf{C}_i^M(D_i^M, D_i^M) \otimes \mathbf{C}^T(D^T, D^T)$.

For the case of arbitrary r , see Bayarri et al. (2005).

3. $\log(\lambda_i^b) \sim N(\eta, 4q^2)$; q needs to be specified, η is given a constant prior.

We note that the Monte Carlo methods described in Appendix B can be extended to provide a simulation-based model fitting technique for the hierarchical model.

In terms of the MCMC, and restricting attention to the CRASH example, the full conditionals of μ_i^M , μ^b , μ , and η are Normal, and hence can be sampled directly.

The common parameters are sampled as a block, using a Metropolis-Hastings step with a proposal similar to the one used in the single barrier analysis, except that now λ^b is excluded from this step as it is not common across the different computer models. To sample from the full conditional of the λ_i^b and τ we follow a similar strategy: we reparametrize in terms of the natural logarithms of these parameters and use a t density proposal centered at the previous iteration and with a scale parameter that needs to be tuned. See Bayarri et al. (2005) for the specifics of how the scales of these proposals were tuned in the CRASH application.